A series of astronomical functions which may be useful. These are all 'user defined functions'. This means that you can paste them into spreadsheets just like the normal functions - see Insert function, you can even use the function wizard. The disadvantage of Excel's 'user defined functions' is that they can only return a single value, and the function cannot alter the properties of the worksheet. Arguments you pass to the VBA functions you define are passed 'by value'. However, VBA defaults to 'passing arguments by reference' when a function is called from another VBA function! This can lead to a function giving a different answer when called in the VBA module compared with when called in the spreadsheet. Use the ByVal keyword to tag arguments you change later in functions. See smoon() for an example. define some numerical constants - these are not accessible in the spreadsheet. Public Const pi As Double = 3.14159265358979 Public Const tpi As Double = 6.28318530717958 Public Const degs As Double = 57.2957795130823 Public Const rads As Double = 1.74532925199433E-02 The trig formulas working in degrees. This just makes the spreadsheet formulas a bit easier to read. DegAtan2() has had the arguments swapped from the Excel order, so the order matches most textbooks Function DegSin(x As Double) As Double DegSin = Sin(rads * x)End Function Function DegCos(x As Double) As Double DegCos = Cos(rads * x)End Function Function DegTan(x As Double) As Double DegTan = Tan(rads * x)End Function Function DegArcsin(x As Double) As Double DegArcsin = degs * Application.Asin(x) End Function Function DegArccos(x As Double) As Double DegArccos = degs * Application.Acos(x) End Function Function DegArctan(x As Double) As Double DegArctan = degs * Atn(x)End Function Function DegAtan2(y As Double, x As Double) As Double this returns the angle in the range 0 to 360instead of -180 to 180 - and swaps the arguments This format matches Meeus and Duffett-Smith DegAtan2 = degs * Application.Atan2(x, y) If DegAtan2 < 0 Then DegAtan2 = DegAtan2 + 360 End Function Private Function range2pi(x)

```
returns an angle x in the range 0 to two pi rads
   This function is not available in the spreadsheet
range2pi = x - tpi * Int(x / tpi)
End Function
Private Function range360(x)
   returns an angle x in the range 0 to 360
   used to prevent the huge values of degrees
    that you get from mean longitude formulas
   this function is private to this module,
   you won't find it in the Function Wizard,
   and you can't use it on a spreadsheet.
   If you want it on the spreadsheet, just remove
   the 'private' keyword above.
range360 = x - 360 * Int(x / 360)
End Function
Function degdecimal(d, m, s)
    converts from dms format to ddd format
    degdecimal = d + m / 60 + s / 3600
End Function
   calander functions. jday and jcentury work on the Julian day numbers.
   day2000 and century2000 work on the days to J2000 to reduce the
   number of significant figures needed
Function jday(year As Integer, month As Integer, day As Integer, hour As Integer, _
min As Integer, sec As Double, Optional greg) As Double
   returns julian day number given date in gregorian calender (greg=1)
   or julian calendar (greg = 0). If greg ommited, then Gregorian is assumed.
   Dim a As Double
   Dim b As Integer
   a = 10000# * year + 100# * month + day
   If (a < -47120101) Then MsgBox "Warning: date too early for algorithm"
    If (IsMissing(greg)) Then greg = 1
    If (month <= 2) Then
       month = month + 12
       year = year -1
   End If
    If (greg = 0) Then
       b = -2 + Fix((year + 4716) / 4) - 1179
   Else
       b = Fix(year / 400) - Fix(year / 100) + Fix(year / 4)
   End If
    a = 365 \# * year + 1720996.5
    jday = a + b + Fix(30.6001 * (month + 1)) + day + (hour + min / 60 + sec / 3600) / 24
End Function
Function jcentury(jd As Double) As Double
    finds how many julian centuries since J2000 given
     the julian day number. Not used below, I just add
    a line into the subroutines which then take days
   before J2000 as the time argument
    jcentury = (jd - 2451545) / 36525
End Function
Function day2000(year As Integer, month As Integer, day As Integer, hour As Integer, _
min As Integer, sec As Double, Optional greg) As Double
   returns days before J2000.0 given date in gregorian calender (greg=1)
   or julian calendar (greg = 0). If you don't provide a value for greg,
   then assumed Gregorian calender
   Dim a As Double
```

```
Dim b As Integer
    If (IsMissing(greg)) Then greg = 1
    a = 10000\# * year + 100\# * month + day
    If (month <= 2) Then
        month = month + 12
        year = year - 1
   End If
    If (greg = 0) Then
        b = -2 + Fix((year + 4716) / 4) - 1179
    Else
        b = Fix(year / 400) - Fix(year / 100) + Fix(year / 4)
    End If
    a = 365 \# * year - 730548.5
    day2000 = a + b + Fix(30.6001 * (month + 1)) + day + (hour + min / 60 + sec / 3600) / 24
End Function
Function century2000(day2000 As Double) As Double
    finds how many julian centuries since J2000 given
    the days before J2000
    century2000 = day2000 / 36525
End Function
   Conversion to and from rectangular and polar coordinates.
   X,Y,Z form a left handed set of axes, and r is the radius vector
   of the point from the origin. Theta is the elevation angle of
   r with the XY plane, and phi is the angle anti-clockwise from the
   X axis and the projection of r in the X,Y plane.
   in astronomical coordinate systems,
   item
            equatorial
                                ecliptic (helio or geo centric)
            celestial pole
                                ecliptic pole
    z
            equatorial plane
                                ecliptic
   x,y
   theta
           declination
                                latitude
   phi
           right ascension
                                longitude
Function rectangular(r As Double, theta As Double, phi As Double, _
 index As Integer) As Double
   takes spherical coordinates in degrees and returns the rectangular
   coordinate shown by index, 1 = x, 2 = y, 3 = z
   x = r.cos(theta).cos(phi)
 1
   y = r.cos(theta).sin(phi)
   z = r.sin(theta)
   Dim r_cos_theta As Double
   r_cos_theta = r * DegCos(theta)
    Select Case index
        Case 1
            rectangular = r_cos_theta * DegCos(phi) 'returns x coord
        Case 2
            rectangular = r_cos_theta * DegSin(phi) 'returns y coord
        Case 3
            rectangular = r * DegSin(theta)
                                                     'returns z coord
    End Select
End Function
Function rlength(x As Double, y As Double, z As Double) As Double
    returns radius vector given the rectangular coords
    rlength = Sqr(x * x + y * y + z * z)
End Function
Function spherical(x As Double, y As Double, z As Double, index As Integer) As Double
    Takes the rectangular coordinates and returns the shperical
    coordinate selected by index -1 = r, 2 = theta, 3 = phi
```

```
astrofns - 4
   r = sqrt(x^*x + y^*y + z^*z)
   tan(phi) = y/x - use atan2 to get in correct quadrant
   tan(theta) = z/sqrt(x*x + y*y) - likewise
   Dim rho As Double
   rho = x * x + y * y
        Select Case index
        Case 1
            spherical = Sqr(rho + z * z) 'returns r
        Case 2
            rho = Sqr(rho)
            spherical = DegArctan(z / rho) 'returns theta
        Case 3
            rho = Sqr(rho)
            spherical = DegAtan2(y, x) 'returns phi
    End Select
End Function
   returns the obliquity of the ecliptic in degrees given the number
   of julian centuries from J2000
   Most textbooks will give the IAU formula for the obliquity of
   the ecliptic below;
   obliquity = 23.43929111 - 46.8150"t - 0.00059"t<sup>2</sup> + 0.001813*t<sup>3</sup>
   as explained in Meeus or Numerical Recipes, it is more efficient and
    accurate to use the nested brackets shown in the function. If you
    multiply the brackets out, they come to the same.
Function obliquity(d As Double) As Double
   Dim t As Double
    t = d / 36525
                   'julian centuries since J2000.0
    obliquity = 23.43929111 - (46.815 + (0.00059 - 0.001813 * t) * t) * t / 3600#
End Function
   functions for converting between equatorial and ecliptic
   geocentric coordinates, both polar and rectangular coords
    Converts geocentric ecliptic coordinates into geocentric equatorial
    coordinates. Expects rectangular coordinates.
Function requatorial(x As Double, y As Double, z As Double, d As Double, _
 index As Integer) As Double
   Dim obl As Double
    obl = obliquity(d)
    Select Case index
        Case 1
            requatorial = x
        Case 2
            requatorial = y * DegCos(obl) - z * DegSin(obl)
        Case 3
            requatorial = y * DegSin(obl) + z * DegCos(obl)
    End Select
End Function
    converts geocentric equatorial coordinates into geocentric ecliptic
    coordinates. Expects rectangular coordinates.
Function recliptic(x As Double, y As Double, z As Double, d As Double, _
 index As Integer) As Double
    Dim obl As Double
    obl = obliquity(d)
```

```
Select Case index
        Case 1
            recliptic = x
        Case 2
            recliptic = y * DegCos(obl) + z * DegSin(obl)
        Case 3
            recliptic = -y * DegSin(obl) + z * DegCos(obl)
    End Select
End Function
    Converts geocentric ecliptic coordinates into geocentric equatorial
    coordinates. Expects spherical coordinates.
Function sequatorial(r As Double, theta As Double, phi As Double, d As Double, _
 index As Integer) As Double
    Dim x As Double, y As Double, z As Double
    x = rectangular(r, theta, phi, 1)
    y = rectangular(r, theta, phi, 2)
    z = rectangular(r, theta, phi, 3)
    sequatorial = spherical(requatorial(x, y, z, d, 1), requatorial(x, y, z, d, 2), _
     requatorial(x, y, z, d, 3), index)
End Function
    Converts geocentric equatorial coordinates into geocentric ecliptic
    coordinates. Expects spherical coordinates.
Function secliptic(r As Double, theta As Double, phi As Double, d As Double, _
 index As Integer) As Double
    Dim x As Double, y As Double, z As Double
    x = rectangular(r, theta, phi, 1)
    y = rectangular(r, theta, phi, 2)
    z = rectangular(r, theta, phi, 3)
    secliptic = spherical(recliptic(x, y, z, d, 1), recliptic(x, y, z, d, 2), _
    recliptic(x, y, z, d, 3), index)
End Function
    precession (approximate formula) from Meeus Algorithms p124
    d1 is the epoch to precess from, d2 is the epoch to precess
    to, and index selects ra or dec. The function takes optional
    arguments dra and ddec to represent the proper motion of a
    star in seconds of arc per year.
    ra and dec must BOTH be in decimal degrees. This formula is
    different to the one elsewhere on the Web site!
Function precess(d1 As Double, d2 As Double, dec As Double, _
                    ra As Double, index As Integer,
                    Optional ddec, Optional dra) As Double
Dim m As Double, n As Double, t As Double
If (IsMissing(dra)) Then dra = 0
If (IsMissing(ddec)) Then ddec = 0
t = d1 / 36525
                        'years since J2000
m = 0.01281233333333 + 0.00000775 * t
n = 0.005567527777778 - 2.3611111111111E-06 * t
t = (d2 - d1) / 365.25
                        'difference in julian _years_, not centuries!
Select Case index
    Case 1
                'dec
        precess = dec + (n * DegCos(ra) + ddec / 3600) * t
                'ra
    Case 2
        precess = ra + (m + n * DegSin(ra) * DegTan(dec) + dra / 3600) * t
End Select
End Function
    The function below returns the geocentric ecliptic coordinates of the sun
    to an accuracy corresponding to about 0.01 degree over
    100 years either side of J2000. Coordinates returned
    in spherical form. From page C24 of the 1996 Astronomical
```

```
Alamanac. Comparing accuracy with Planeph, a DOS ephemeris
   by Chapront, we get;
   Sun error
                       RA sec DEC arcsec
                                   8.9
   Max within 3 year
                        0.6
   Min within 3 year
                        -2.1
                                   -8.2
   Max within 10 year 0.6
                                   10.9
   Min within 10 year
                        -2.6
                                   -12.5
   Max within 50 year 1.0
                                   16.8
   Min within 50 year
                        -2.9
                                   -16.1
   Error = C24 low precision method - Planeph
   Note: Planeph was set to give output referred to mean
          ecliptic and equinox of date.
   The accuracy of this routine is good enough for sunrise
   and shadow direction calculations, and for referring
   low precision planetary and comet positions to the Earth,
   but is no good for accurate coordinate conversion or
    for eclipse or occultation use.
    Coordinates are referred to the ecliptic and mean equinox of date
Function ssun(d As Double, index As Integer) As Double
   Dim g As Double
   Dim 1 As Double
    g = range360(357.528 + 0.9856003 * d)
    1 = range360(280.461 + 0.9856474 * d)
    Select Case index
        Case 1
            ' radius vector of Sun
            ssun = 1.00014 - 0.01671 * DegCos(g) - 0.00014 * DegCos(2 * g)
        Case 2
           ssun = 0
                       'ecliptic latitude of Sun is zero to very good accuracy
        Case 3
            'longitude of Sun
            ssun = range360(1 + 1.915 * DegSin(g) + 0.02 * DegSin(2 * g))
   End Select
End Function
   returns the geocentric ecliptic coordinates of the sun
   to an accuracy corresponding to about 0.01 degree over
   100 years either side of J2000. Assumes ssun() exists.
   rectangular form is easier for converting positions from helio-
    centric to geocentric, but beware low accuracy (roughly 0.01 degree)
    of values
Function rsun(d As Double, index As Integer) As Double
   Dim x As Double
    Dim y As Double
    Dim z As Double
    rsun = rectangular(ssun(d, 1), ssun(d, 2), ssun(d, 3), index)
End Function
    sun() returns the geocentric ra and dec and radius vector
    of the moon - calls smoon three times, and sequatorial
    three times - sequatorial calls rectangular three times
    each!
Function sun(d As Double, index As Integer) As Double
    sun = sequatorial(ssun(d, 1), ssun(d, 2), ssun(d, 3), d, index)
End Function
    The function below implements Paul Schlyter's simplification
ı.
    of van Flandern and Pulkkinen's method for finding the geocentric
```

```
astrofns - 7
    ecliptic positions of the Moon to an accuracy of about 1 to 4 arcmin.
    I can probably reduce the number of variables, and there must
   be a quicker way of declaring variables!
    The VBA trig functions have been used throughout for speed,
   note how the atan function returns values in domain -pi to pi
Function smoon(ByVal d As Double, index As Integer) As Double
    Dim Nm As Double, im As Double, wm As Double, am As Double, ecm As Double, _
    Mm As Double, em As Double, Ms As Double, ws As Double, xv As Double, _
   yv As Double, vm As Double, rm As Double, x As Double, y As Double, _ z As Double, lon As Double, lat As Double, l<br/>s As Double, lm As Double, _
    dm As Double, F As Double, dlong As Double, dlat As Double
        Paul's routine uses a slightly different definition of
        the day number - I adjust for it below. Remember that VBA
        defaults to 'pass by reference' so this change in d
        will be visible to other functions unless you set d to 'ByVal'
        to force it to be passed by value!
    d = d + 1.5
    .
       moon elements
   Nm = range360(125.1228 - 0.0529538083 * d) * rads
    im = 5.1454 * rads
    wm = range360(318.0634 + 0.1643573223 * d) * rads
    am = 60.2666 '(Earth radii)
    ecm = 0.0549
   Mm = range360(115.3654 + 13.0649929509 * d) * rads
       position of Moon
    em = Mm + ecm * Sin(Mm) * (1# + ecm * Cos(Mm))
   xv = am * (Cos(em) - ecm)
   yv = am * (Sqr(1# - ecm * ecm) * Sin(em))
    vm = Application.Atan2(xv, yv)
        If vm < 0 Then vm = tpi + vm
    .
   rm = Sqr(xv * xv + yv * yv)
   x = rm * (Cos(Nm) * Cos(vm + wm) - Sin(Nm) * Sin(vm + wm) * Cos(im))
   y = rm * (Sin(Nm) * Cos(vm + wm) + Cos(Nm) * Sin(vm + wm) * Cos(im))
    z = rm * (Sin(vm + wm) * Sin(im))
       moons geocentric long and lat
   lon = Application.Atan2(x, y)
    If lon < 0 Then lon = tpi + lon
    lat = Atn(z / Sqr(x * x + y * y))
      mean longitude of sun
   ws = range360(282.9404 + 0.0000470935 * d) * rads
   Ms = range360(356.047 + 0.9856002585 * d) * rads
      perturbations
    1
      first calculate arguments below,
    'Ms, Mm
                        Mean Anomaly of the Sun and the Moon
                        Longitude of the Moon's node
    ' Nm
                       Argument of perihelion for the Sun and the Moon
    'ws, wm
                      'Mean Longitude of the Sun (Ns=0)
    ls = Ms + ws
    lm = Mm + wm + Nm 'Mean longitude of the Moon
   dm = lm - ls
                        'Mean elongation of the Moon
                       'Argument of latitude for the Moon
   F = lm - Nm
    ' then add the following terms to the longitude
    ' note amplitudes are in degrees, convert at end
    Select Case index
        Case 1 ' distance terms earth radii
            rm = rm - 0.58 * Cos(Mm - 2 * dm)
            rm = rm - 0.46 * Cos(2 * dm)
            smoon = rm
        Case 2 ' latitude terms
            dlat = -0.173 * Sin(F - 2 * dm)
            dlat = dlat - 0.055 * Sin(Mm - F - 2 * dm)
            dlat = dlat - 0.046 * Sin(Mm + F - 2 * dm)
            dlat = dlat + 0.033 * Sin(F + 2 * dm)
            dlat = dlat + 0.017 * Sin(2 * Mm + F)
            smoon = lat * degs + dlat
        Case 3 ' longitude terms
```

```
dlon = -1.274 * Sin(Mm - 2 * dm)
                                                   '(the Evection)
            dlon = dlon + 0.658 * Sin(2 * dm)
                                                   '(the Variation)
            dlon = dlon - 0.186 * Sin(Ms)
                                                    '(the Yearly Equation)
            dlon = dlon - 0.059 * Sin(2 * Mm - 2 * dm)
            dlon = dlon - 0.057 * Sin(Mm - 2 * dm + Ms)
            dlon = dlon + 0.053 * Sin(Mm + 2 * dm)
            dlon = dlon + 0.046 * Sin(2 * dm - Ms)
            dlon = dlon + 0.041 * Sin(Mm - Ms)
            dlon = dlon - 0.035 * Sin(dm)
                                                     '(the Parallactic Equation)
            dlon = dlon - 0.031 * Sin(Mm + Ms)
            dlon = dlon - 0.015 * Sin(2 * F - 2 * dm)
            dlon = dlon + 0.011 * Sin(Mm - 4 * dm)
            smoon = lon * degs + dlon
    End Select
End Function
    rmoon uses smoon to return the geocentric ecliptic rectangular coordinates
    of the moon to lowish accuracy.
Function rmoon(d As Double, index As Integer) As Double
    rmoon = rectangular(smoon(d, 1), smoon(d, 2), smoon(d, 3), index)
End Function
    moon() returns the geocentric ra and dec and radius vector
    of the moon - calls smoon three times, and sequatorial
    three times - sequatorial calls rectangular three times
    each!
Function moon(d As Double, index As Integer) As Double
Dim nigel As Double
    If index = 4 Then
        nigel = smoon(d, 2)
        moon = d
    Else
       moon = sequatorial(smoon(d, 1), smoon(d, 2), smoon(d, 3), d, index)
    End If
End Function
    Solutions of the kepler equation using a Newton's method approach.
    See Meeus or Duffett-Smith (calculator)
Function kepler(m As Double, ecc As Double, Optional eps)
    solves the equation e - ecc*sin(e) = m for e given an m
    returns the value of the 'true anomaly' in rads
    m - the 'mean anomaly' in rads
    ecc - the eccentricity of the orbit
    eps - the precision parameter - solution will be
          within 10<sup>-</sup>-eps of the true value.
          don't set eps above 14, as convergence
          can't be guaranteed. If not specified, then
          taken as 10<sup>-8</sup> or 10 nano radians!
    Dim delta As Double, e As Double, v As Double
    e = m
                        'first guess
                        'set delta equal to a dummy value
    delta = 0.05
    If (IsMissing(eps)) Then eps = 8
                                                     'if no eps then assume 10^-8
    Do While Abs(delta) >= 10 ^ -eps
                                                     'converged?
        delta = e - ecc * Sin(e) - m
                                                     'new error
        e = e - delta / (1 - ecc * Cos(e))
                                                     'corrected guess
    Loop
    v = 2 * Atn(((1 + ecc) / (1 - ecc)) ^ 0.5 * Tan(0.5 * e))
    If v < 0 Then v = v + tpi
    kepler = v
End Function
ı.
    The functions below return the heliocentric ecliptic coordinates
```

```
of the planets to an accuracy of a few minutes of arc. The coordinates
    are referred to the equinox of J2000.0
   The functions use a simple Kepler ellipse, but with
   mean elements which change slightly with the time since
   J2000. See 'Explanatory supplement to the Astronomical
   Almanac' 1992, page 316, table 5.8.1. Worst case errors
   over the period 1800 - 2050 AD in arcsec are below;
                            Dec
                     Ra
   Mercury
                     20"
                             5 "
                     20"
                              5 "
   Venus
   Earth
                     20"
                              5"
                             30"
   Mars
                    25"
                   300"
   Jupiter
                           100"
                   600"
   Saturn
                            200"
                   60"
                           25"
   Uranus
                            20"
                    40"
   Neptune
                    40"
   Pluto
                            10"
   The rplanet() function returns the ecliptic heliocentric coordinates
    of each of the major planets. You select the planet you want using
    pnumber, and the coordinate you want with index as usual.
Function rplanet(d As Double, pnumber As Integer, index As Integer) As Double
    Dim x As Double, y As Double, z As Double, v As Double, m As Double, _
    i As Double, o As Double, p As Double, a As Double, e As Double, _
    l As Double, r As Double
       get elements of the planet
    element i, o, p, a, e, l, d, pnumber
       position of planet in its orbit
   m = range2pi(l - p)
   v = kepler(m, e, 8)
   r = a * (1 - e * e) / (1 + e * Cos(v))
    1
       heliocentric rectangular coordinates of planet
    Select Case index
    Case 1
               'x coordinate
       rplanet = r * (Cos(o) * Cos(v + p - o) - Sin(o) * Sin(v + p - o) * Cos(i))
    Case 2
               'y coordinate
       rplanet = r * (Sin(o) * Cos(v + p - o) + Cos(o) * Sin(v + p - o) * Cos(i))
               'z coordinate
    Case 3
        rplanet = r * (Sin(v + p - o) * Sin(i))
    End Select
End Function
    The planet() function returns the equatorial geocentric coordinates
    of each of the major planets. You select the planet you want using
   pnumber, and the coordinate you want with index as usual. Code is
   duplicated from rplanet() to reduce the number of calls to kepler()
Function planet(d As Double, pnumber As Integer, index As Integer) As Double
   Dim x As Double, y As Double, z As Double, v As Double, m As Double, _
    i As Double, o As Double, p As Double, a As Double, e As Double, _
    1 As Double, r As Double, xe As Double, ye As Double, ze As Double, _
    sl As Double, si As Double, so As Double, cl As Double, ci As Double, _
    co As Double
```

```
' elements of planet - select from the values
```

```
astrofns - 10
```

```
element i, o, p, a, e, l, d, pnumber
.
   position of planet in its orbit
m = range2pi(l - p)
v = kepler(m, e, 8)
r = a * (1 - e * e) / (1 + e * Cos(v))
    heliocentric rectangular coordinates of planet
s1 = Sin(v + p - o)
si = Sin(i)
so = Sin(o)
c1 = Cos(v + p - o)
ci = Cos(i)
co = Cos(o)
x = r * (co * cl - so * sl * ci)
y = r * (so * c1 + co * s1 * ci)
z = r * (s1 * si)
1
    elements of earth (reusing variables)
element i, o, p, a, e, l, d, 3
.
    position of earth in its orbit
m = range2pi(l - p)
v = kepler(m, e, 8)
r = a * (1 - e * e) / (1 + e * Cos(v))
    heliocentric rectangular coordinates of earth
s1 = Sin(v + p - o)
si = Sin(i)
so = Sin(o)
c1 = Cos(v + p - o)
ci = Cos(i)
co = Cos(o)
xe = r * (co * cl - so * sl * ci)
ye = r * (so * c1 + co * s1 * ci)
ze = r * (sl * si)
ı.
    convert to geocentric rectangular coordinates
1
x = x - xe
y = y - ye
   z = z
    rotate around x axis from ecliptic to equatorial coords
xe = x
ye = y * Cos(ecl) - z * Sin(ecl)
ze = y * Sin(ecl) + z * Cos(ecl)
1
    find the RA and DEC from the rectangular equatorial coords
Select Case index
Case 3
    ' RA in degrees
    planet = Application.Atan2(xe, ye) * degs
    If planet < 0 Then planet = 360 + planet
Case 2
    ' DEC in degrees
   planet = Atn(ze / Sqr(xe * xe + ye * ye)) * degs
Case 1
    ' Radius vector in au
```

```
planet = Sqr(xe * xe + ye * ye + ze * ze)
   End Select
End Function
   The subroutine below replaces the values of i,o,p,a,e,L
   with the values for the planet selected by pnum. You could
   always add planet like objects, but watch the value of
   the inclination i. The method used in planet is only
   good for orbits 'near' the ecliptic.
   This is an example of the Visual Basic default 'passing by
   reference'
Sub element(i As Double, o As Double, p As Double, _
            a As Double, e As Double, l As Double,
            ByVal d As Double, ByVal pnum As Integer)
   Select Case pnum
       Case 1
                        'mercury
            i = (7.00487 - 0.000000178797 * d) * rads
            o = (48.33167 - 0.0000033942 * d) * rads
           p = (77.45645 + 0.00000436208 * d) * rads
            a = 0.38709893 + 1.80698E-11 * d
            e = 0.20563069 + 0.00000000691855 * d
            l = range2pi(rads * (252.25084 + 4.092338796 * d))
       Case 2
                        'venus
            i = (3.39471 - 0.000000217507 * d) * rads
            o = (76.68069 - 0.0000075815 * d) * rads
            p = (131.53298 - 0.00000827439 * d) * rads
            a = 0.72333199 + 2.51882E-11 * d
            e = 0.00677323 - 0.0000000135195 * d
            l = range2pi(rads * (181.97973 + 1.602130474 * d))
       Case 3
                        'earth
            i = (0.00005 - 0.000000356985 * d) * rads
            o = (-11.26064 - 0.00013863 * d) * rads
            p = (102.94719 + 0.00000911309 * d) * rads
            a = 1.00000011 - 1.36893E-12 * d
            e = 0.01671022 - 0.0000000104148 * d
            l = range2pi(rads * (100.46435 + 0.985609101 * d))
       Case 4
                        'mars
            i = (1.85061 - 0.000000193703 * d) * rads
            o = (49.57854 - 0.0000077587 * d) * rads
            p = (336.04084 + 0.00001187 * d) * rads
            a = 1.52366231 - 0.000000001977 * d
            e = 0.09341233 - 0.0000000325859 * d
            1 = range2pi(rads * (355.45332 + 0.524033035 * d))
        Case 5
                        'jupiter
            i = (1.3053 - 0.000000315613 * d) * rads
            o = (100.55615 + 0.00000925675 * d) * rads
            p = (14.75385 + 0.00000638779 * d) * rads
            a = 5.20336301 + 0.0000000166289 * d
            e = 0.04839266 - 0.0000000352635 * d
            l = range2pi(rads * (34.40438 + 0.083086762 * d))
       Case 6
                        'saturn
            i = (2.48446 + 0.000000464674 * d) * rads
            o = (113.71504 - 0.0000121 * d) * rads
            p = (92.43194 - 0.0000148216 * d) * rads
            a = 9.53707032 - 0.000000825544 * d
            e = 0.0541506 - 0.0000000100649 * d
            l = range2pi(rads * (49.94432 + 0.033470629 * d))
       Case 7
                        'uranus
            i = (0.76986 - 0.000000158947 * d) * rads
            o = (74.22988 + 0.0000127873 * d) * rads
            p = (170.96424 + 0.0000099822 * d) * rads
            a = 19.19126393 + 0.0000000416222 * d
            e = 0.04716771 - 0.0000000524298 * d
            l = range2pi(rads * (313.23218 + 0.011731294 * d))
       Case 8
                        'neptune
            i = (1.76917 - 0.000000276827 * d) * rads
```

```
o = (131.72169 - 0.0000011503 * d) * rads
p = (44.97135 - 0.00000642201 * d) * rads
a = 30.06896348 - 0.0000000342768 * d
e = 0.00858587 + 0.00000000688296 * d
l = range2pi(rads * (304.88003 + 0.0059810572 * d)))
Case 9 'pluto
i = (17.14175 + 0.000000841889 * d) * rads
o = (110.30347 - 0.0000002839 * d) * rads
p = (224.06676 - 0.00000100578 * d) * rads
a = 39.48168677 - 0.0000000210574 * d
e = 0.24880766 + 0.0000000177002 * d
l = range2pi(rads * (238.92881 + 3.97557152635181E-03 * d)))
End Select
End Sub
```